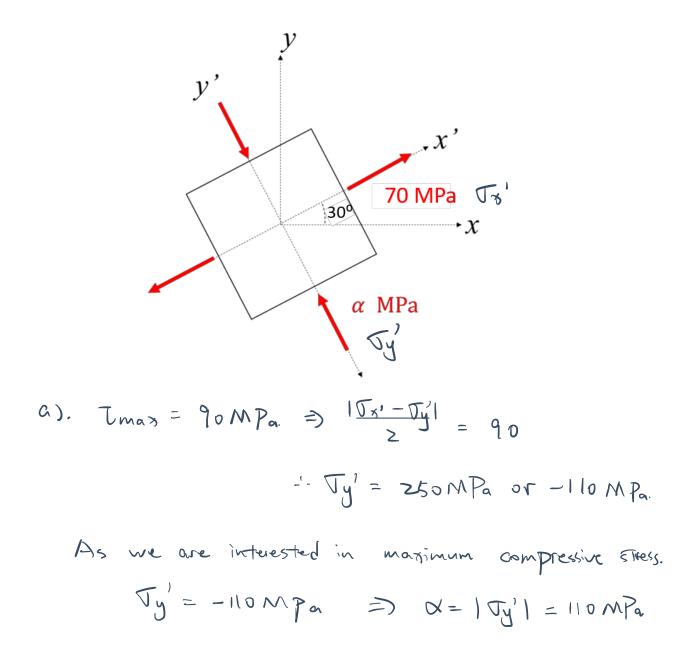
## Problem 10.1 (10 points)

The stress element shown below represents the state of stress measured along the x'y' axis in a component loaded under plane stress. No information is known about the stress  $\alpha$  except that it is compressive.

- a) Determine the magnitude of the maximum compressive normal stress  $\alpha$  that can be applied, if the component is made of a material which can withstand a maximum in-plane shear stress of 90 MPa.
- b) Determine the stress components when the element is oriented along the x-y axes.
- c) Draw a stress element oriented along the maximum in-plane shear stress directions. (Show the angle of this rotated element with respect to the axis x')



b). 
$$T_{x} = \frac{T_{x}' + T_{y}'}{2} + \frac{T_{x}' - T_{y}'}{2} \cos 2\theta + \frac{T_{x}' - T_{y}'}{2} \sin 2\theta$$

$$= 25 \text{ MPa}$$

$$T_{y} = \frac{T_{x}' + T_{y}'}{2} - \frac{T_{x}' - T_{y}'}{2} \cos 2\theta - \frac{T_{x}' - T_{y}'}{2} \sin 2\theta$$

$$= -65 \text{ MPa}$$

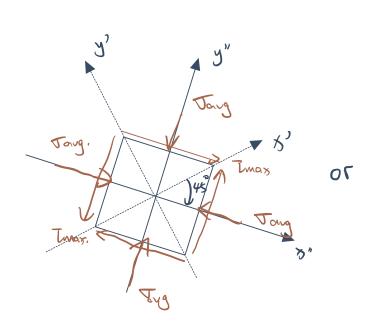
$$T_{xy} = -\left(\frac{T_{x}' - T_{y}'}{2}\right) \sin 2\theta + \frac{T_{x}' - T_{y}'}{2} \cos 2\theta$$

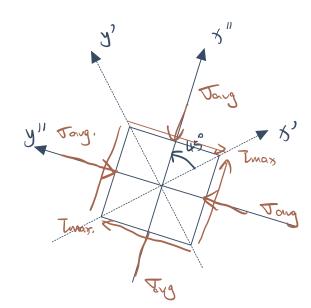
$$= 45 \sqrt{3} \text{ MPa}.$$

C) 
$$\beta = \pm 45^{\circ}$$
 with respect to  $5'$ , for a positive Tmax.  
 $T_{5''}y'' = T_{max} = -\frac{T_{7'} - T_{7'}}{2} \sin 2\beta + T_{5''}\cos 2\beta$ 

$$= 90 \text{ MPa} \implies \beta = -45^{\circ}$$

$$T_{7''} = T_{7''} = T_{ave} = \frac{T_{7'} + T_{7'}}{2} = -20 \text{ MPa}.$$

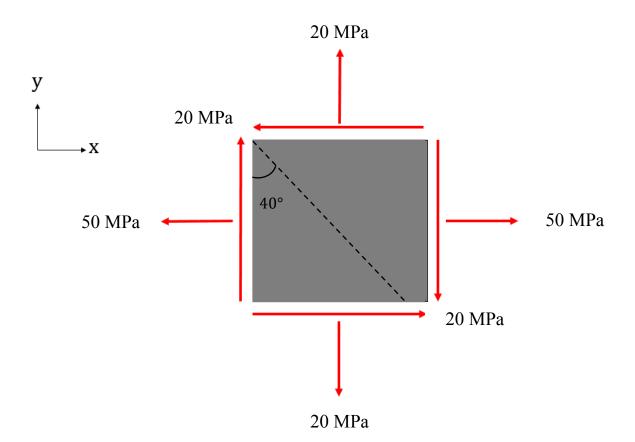




## Problem 10.2 (10 points)

For the state of plane stress shown in the figure:

- a) Draw the Mohr's circle and indicate the points that represent stresses on face x and on face y.
- b) Using the Mohr's circle, determine the normal and shear stress on the inclined plane shown in the figure and label this point as N on the Mohr's circle.



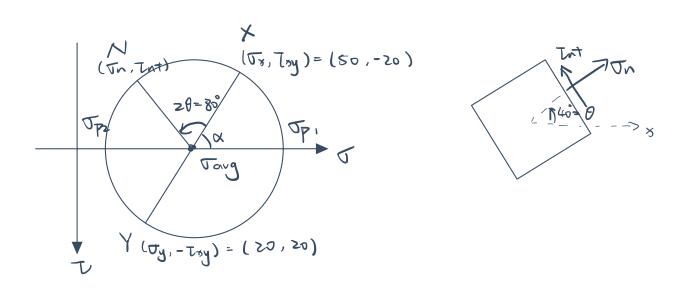
$$\nabla_{x} = 5 \circ MP_{\alpha}$$

$$\nabla_{y} = 20 MP_{\alpha}$$

$$\nabla_{y} = -20 MP_{\alpha}$$

$$R = \sqrt{\left(\frac{\sigma_{x} - \sigma_{y}}{2}\right)^{2} + \tau_{xy}^{2}}$$

$$\Rightarrow \nabla_{xy} = -25 MP_{\alpha}$$

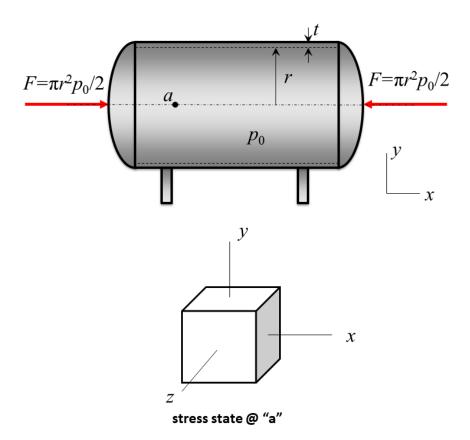


$$tan x = \frac{20}{50-35}$$
  $\Rightarrow x = 0.927 \, rad = 53.13°$ 
 $N = (R \cos (\alpha + 2\theta) + \sqrt{\alpha} , -R \sin (\alpha + 2\theta))$ 
 $= (17.9 \, MPa, -18.25 \, MPa).$ 
 $\sqrt{17.9 \, MPa}$ 
 $\sqrt{17.9 \, MPa}$ 
 $\sqrt{17.9 \, MPa}$ 

## Problem 10.3 (10 points)

The cylindrical pressure vessel shown below has an inner radius of r and a wall thickness of t, with semi-spherical end caps (t /r << 1). The pressure vessel contains a gas that is under a pressure of  $p_0$ . A pair of compressive loads (force)  $F=\pi r^2 p_0/2$  is applied on the end caps of the vessel.

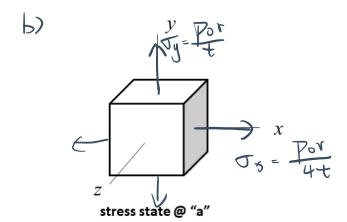
- a) Determine the state of stress at point "a" on the front of the vessel.
- b) Show the components of stress on the stress element provided.
- c) Make an accurate drawing of the three Mohr's circles for this state of stress.
- d) The allowable tensile stress is 100 MPa and the allowable shear stress is 40 MPa. What is the minimum wall thickness of the vessel? At this step, use  $p_0 = 1$  MPa, r = 0.5 m in your analysis.

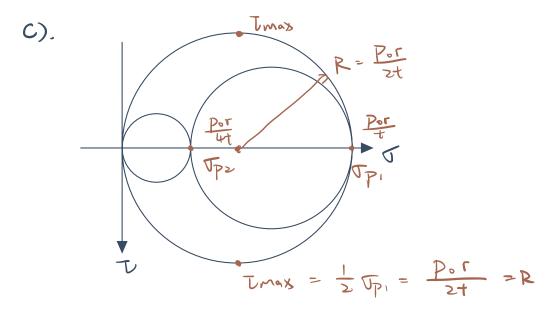


a). 
$$abla a = \frac{Por}{2t}$$

$$= \frac{Por}{2t} - \frac{Tor^{2}Po/2}{2Tort}$$

$$= \frac{Por}{4t}$$



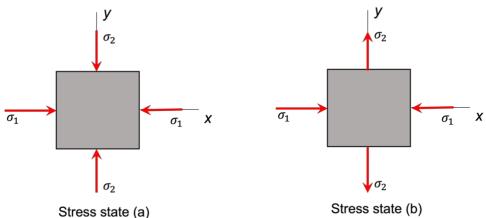


d) 0 
$$\sqrt{p_1} = \sqrt{max} = 100 MPa = \frac{Por}{t}$$
  $\Rightarrow$   $t \ge \frac{Por}{100} = 5 mm$ 

① 
$$T_{max} = 40MP_{a} = \frac{P_{o}r}{24}$$
  $\Rightarrow t > \frac{P_{o}r}{80} = 6.25mm$ 

## Problem 10.4 (5 points)

Consider stress states (a) and (b) shown above, with  $|\sigma_1| > |\sigma_2|$ . Let  $(|\tau|_{max,abs})_a$  and  $(|\tau|_{max,abs})_b$  represent the absolute maximum shear stress corresponding to stress states (a) and (b), respectively. Choose the response below that describes the relative sizes of these stresses.



i. 
$$(|\tau|_{max,abs})_a > (|\tau|_{max,abs})_b$$

ii. 
$$(|\tau|_{max,abs})_a = (|\tau|_{max,abs})_b$$

(iii.) 
$$(|\tau|_{max,abs})_a < (|\tau|_{max,abs})_b$$