

# ***MECHANICS OF MATERIALS***

**Fall 2023**

**ME 323- 005**

**Instructor: Shubhra Bansal**

**Lecture 2: Normal Stress**

**Extensional Strain**

**Material Properties**

# Common States of Stress

- **Simple tension:** cable



$A_0$  = cross-sectional area (when unloaded)

$\sigma = \frac{F}{A_0}$

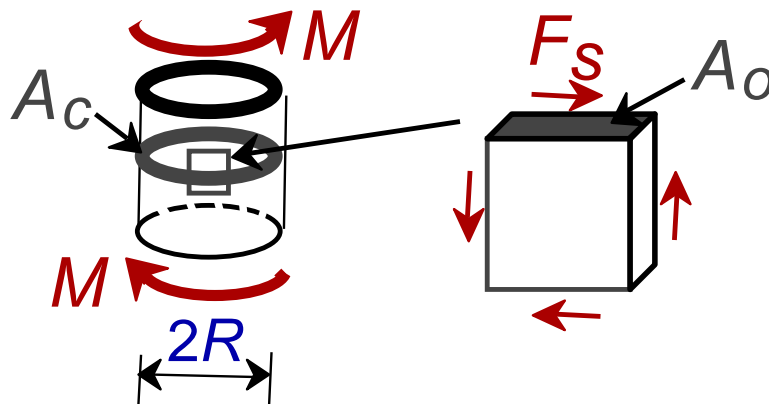
$\sigma \leftarrow \blacksquare \rightarrow \sigma$

$N/m^2$   
 $Pa$

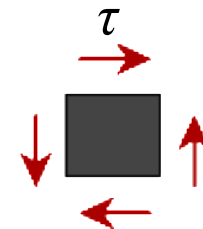


Ski lift (photo courtesy P.M. Anderson)

- **Torsion** (a form of shear): drive shaft



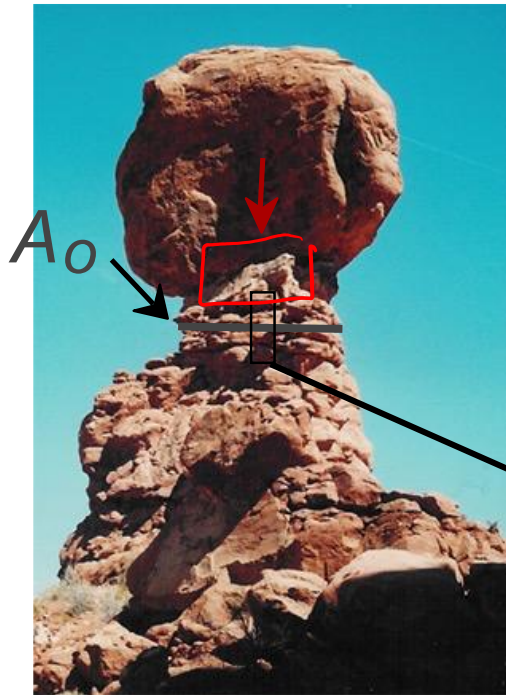
$$\tau = \frac{F_s}{A_0}$$



Note:  $\tau = M/A_c R$  here.

# Common States of Stress

- **Simple** compression:



Balanced Rock, Arches National Park  
(photo courtesy P.M. Anderson)



Canyon Bridge, Los Alamos, NM  
(photo courtesy P.M. Anderson)

$$\sigma = \frac{F}{A_o}$$



Note: compressive structure member  
( $\sigma < 0$  here).

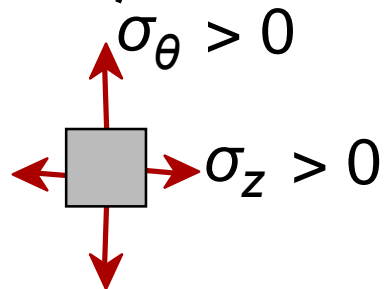
$N/m^2$   
 $Pa$ ,  $MPa$ ,  $GPa$   
 $10^6$ ,  $10^9$

# Common States of Stress

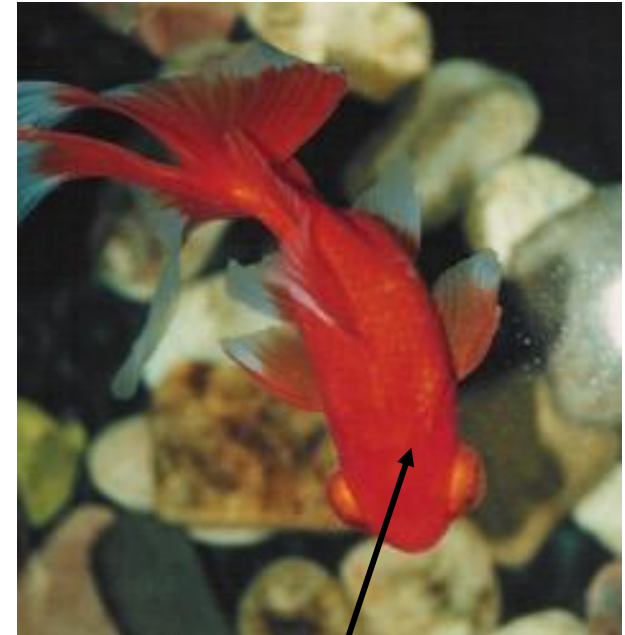
- **Bi-axial** tension:



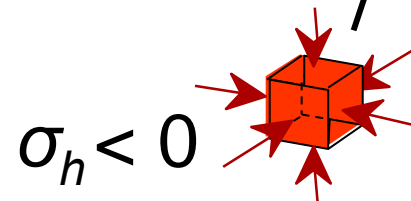
Pressurized tank  
(photo courtesy  
P.M. Anderson)



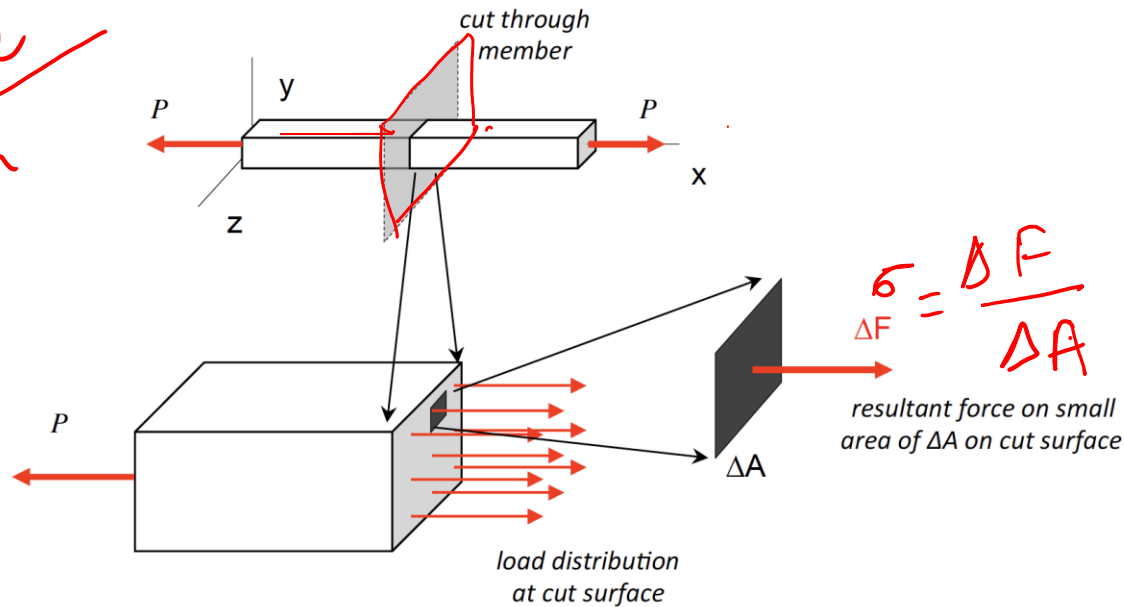
- **Hydrostatic** compression:



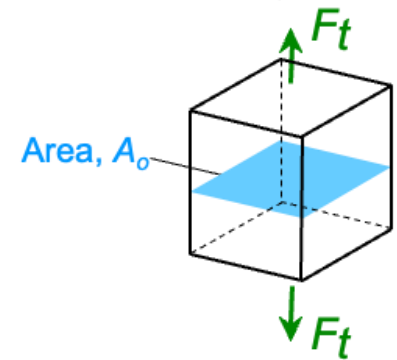
Fish under water  
(photo courtesy  
P.M. Anderson)



# Normal or Axial Stress



- **Tensile stress,  $\sigma$ :**



$$\sigma = \frac{F_t}{A_o} = \frac{\text{lb}_f}{\text{in}^2} \text{ or } \frac{\text{N}}{\text{m}^2}$$

original cross-sectional area before loading

$$\underline{\underline{\sigma_x = \text{normal stress} = \lim_{\Delta A \rightarrow 0} \left( \frac{\Delta F}{\Delta A} \right) = \frac{dF}{dA} \Rightarrow dF = \sigma_x dA}}$$

Average normal stress  $(\sigma_x)_{ave} = \frac{1}{A} \int_{\text{area}} \sigma_x dA$

$$(\sigma_x)_{ave} = \frac{P}{A}$$

$$\sigma_{ave} = \frac{1}{A} \int_{\text{area}} \frac{dF}{dA} \cdot dA$$

- Assuming axial load is at the centroidal position ✓
- Material is homogeneous and isotropic ✓
- Tensile (+) and compressive (-)

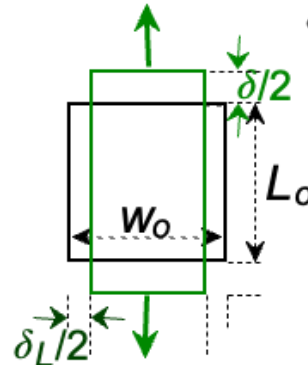
# Axial Strain

- **Tensile strain:**

*Axial*

$$e = \frac{\delta}{L_0}$$

- **Lateral strain:**



$$e_L = -\frac{\delta_L}{W_0}$$

*$\nu = \frac{\text{lateral}}{\text{axial}}$*

*strain =  $\frac{\text{change in dimension}}{\text{original dimension}}$*

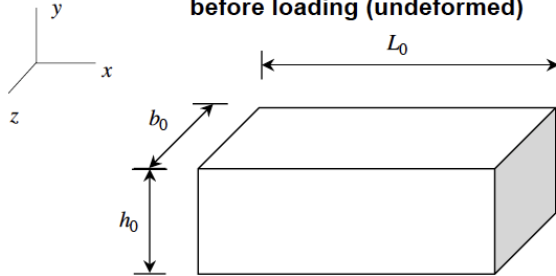
$$= \frac{\Delta L}{L_0} \quad \text{or} \quad \frac{\Delta d}{d_0}$$

$$\epsilon_x = \text{strain in } x \text{ (axial) direction} = \frac{L - L_0}{L_0} = \frac{\Delta L}{L_0} \quad (\text{elongation in } x\text{-direction})$$

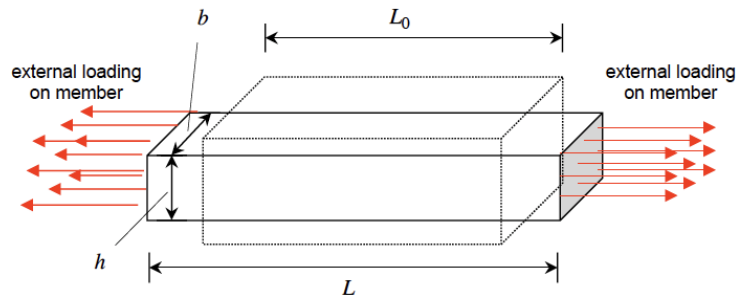
$$\epsilon_y = \text{strain in } y \text{ direction} = \frac{h - h_0}{h_0} = \frac{\Delta h}{h_0} \quad (\text{contraction in } y\text{-direction})$$

$$\epsilon_z = \text{strain in } z \text{ direction} = \frac{b - b_0}{b_0} = \frac{\Delta b}{b_0} \quad (\text{contraction in } z\text{-direction})$$

before loading (undeformed)

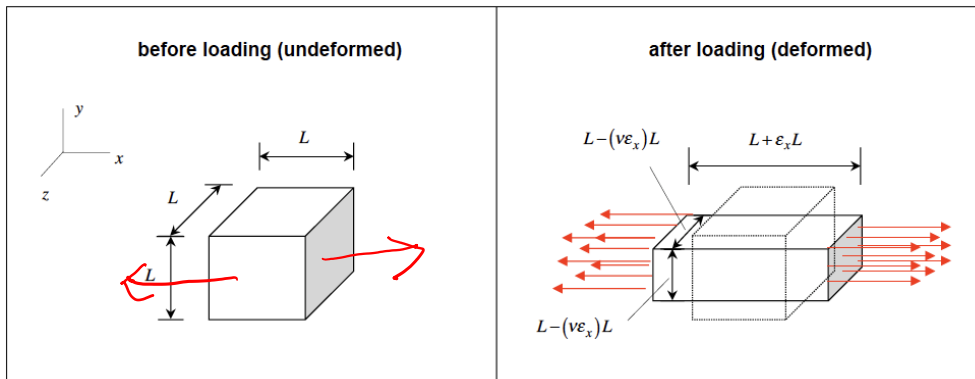


after loading (deformed)



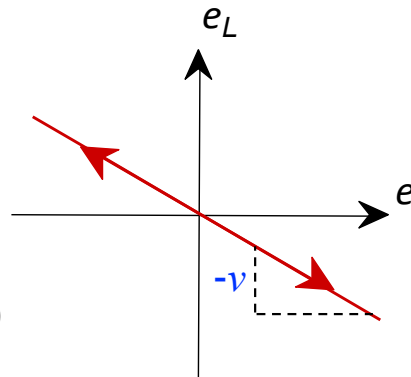


# Poisson's Ratio ( $\nu$ )



$$\nu = -\frac{e_L}{e}$$

metals:  $\nu \sim 0.33$   
 ceramics:  $\nu \sim 0.25$   
 polymers:  $\nu \sim 0.40$

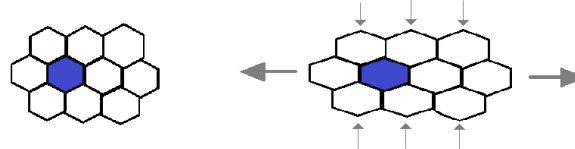


Negative Poisson's ratio

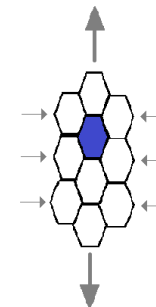
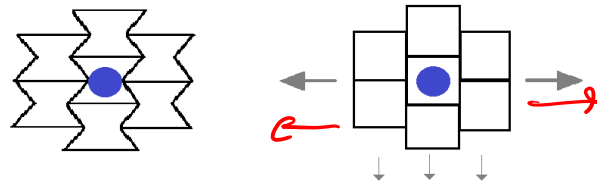
Material	Poisson's ratio
rubber	0.4999 [5]
gold	0.42–0.44
saturated clay	0.40–0.49
magnesium	0.252–0.289
titanium	0.265–0.34
copper	0.33
aluminium-alloy	0.32
clay	0.30–0.45
stainless steel	0.30–0.31
steel	0.27–0.30
cast iron	0.21–0.26
sand	0.20–0.45
concrete	0.1–0.2
glass	0.18–0.3
foam	0.10–0.50
cork	0.0

Auxetic Materials- negative Poisson's Ratio

NORMALE MATERIALIEN



AUXETISCHE MATERIALIEN

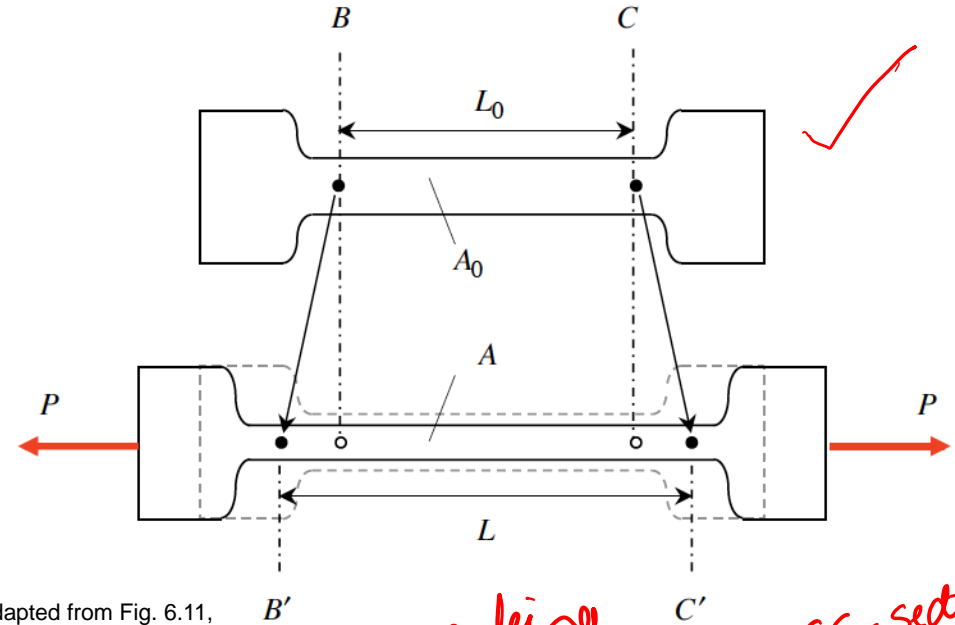


In footwear, auxetic design allows the sole to expand in size while walking or running, thereby increasing flexibility.

# Mechanical Properties of Materials

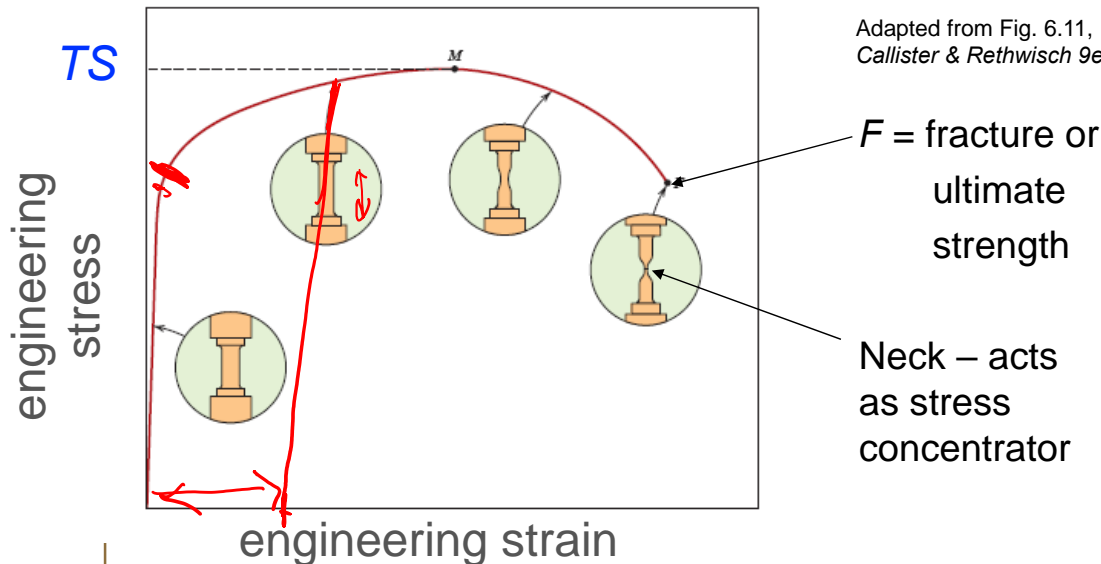
$$\sigma = \frac{P}{A_0}$$

$$\epsilon = \frac{\Delta L}{L_0} = \frac{L - L_0}{L_0}$$



Adapted from Fig. 6.11,  
Callister & Rethwisch 9e.

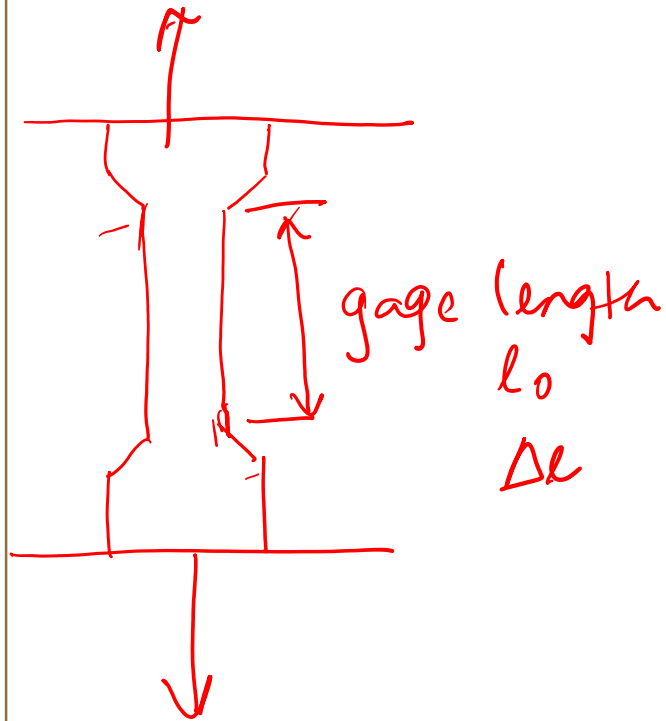
necking → cross-section  
area decreases



- **Metals:** occurs when noticeable necking starts.
- **Polymers:** occurs when polymer backbone chains are aligned and about to break.



# Material Properties mechanical

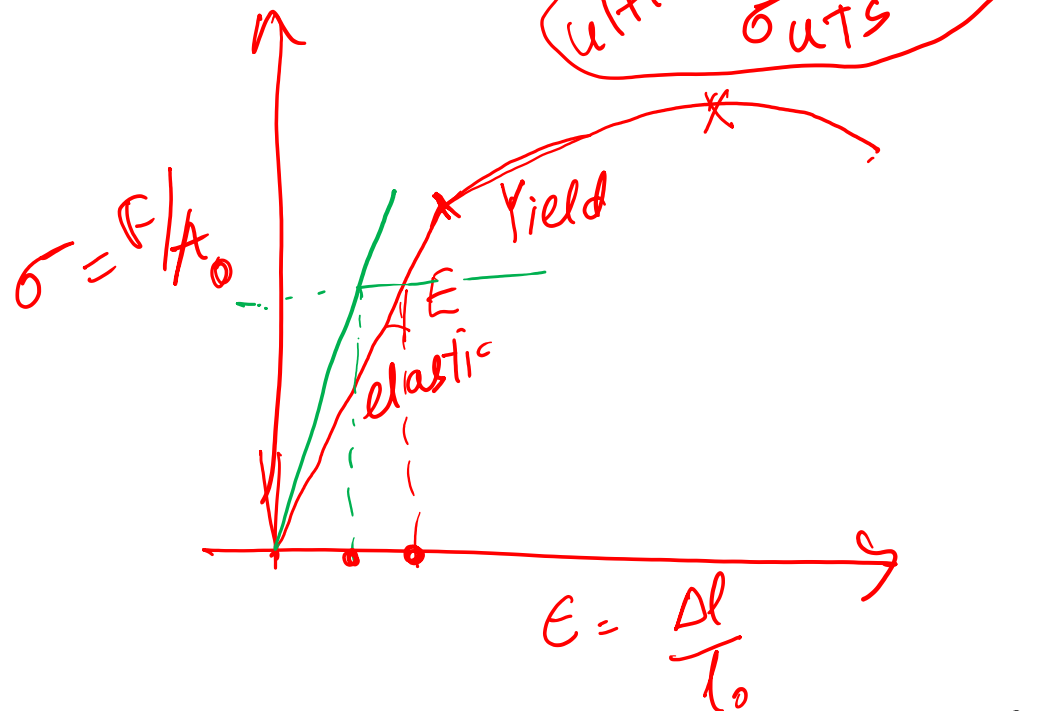


$$\sigma = E \epsilon$$

Hook's Law

elastic  $\rightarrow$  reversible  
plastic  $\rightarrow$  permanent

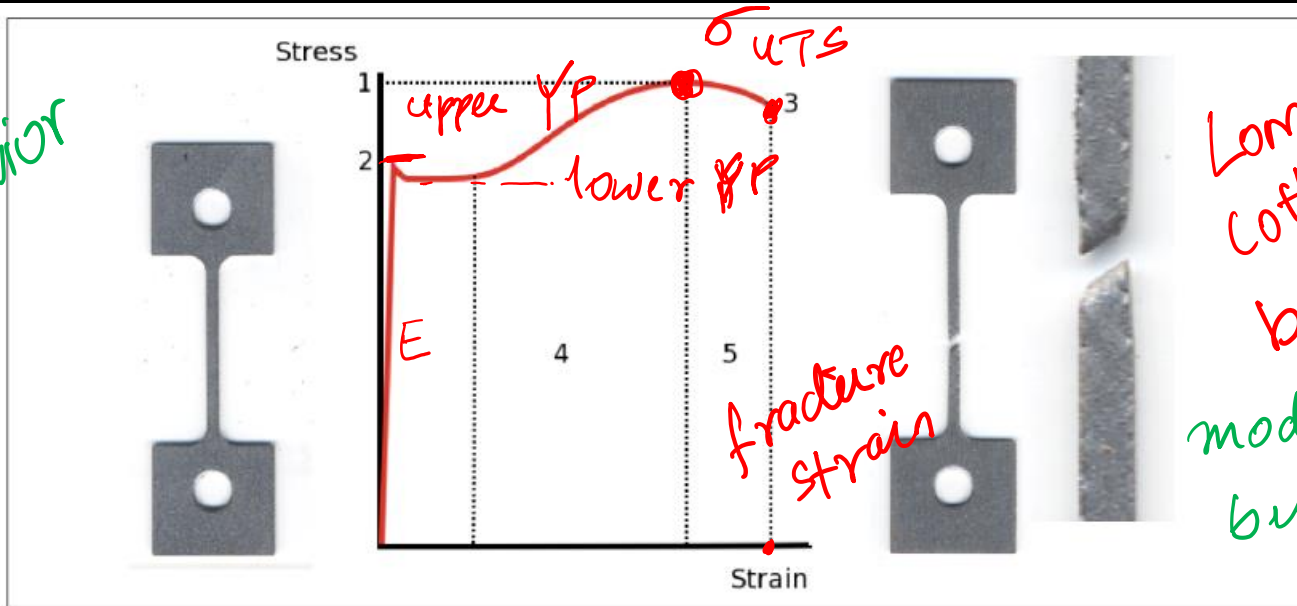
elastic modulus  $\rightarrow$  stiffness  
Young's modulus ( $E$ )



# Stress-Strain Curves

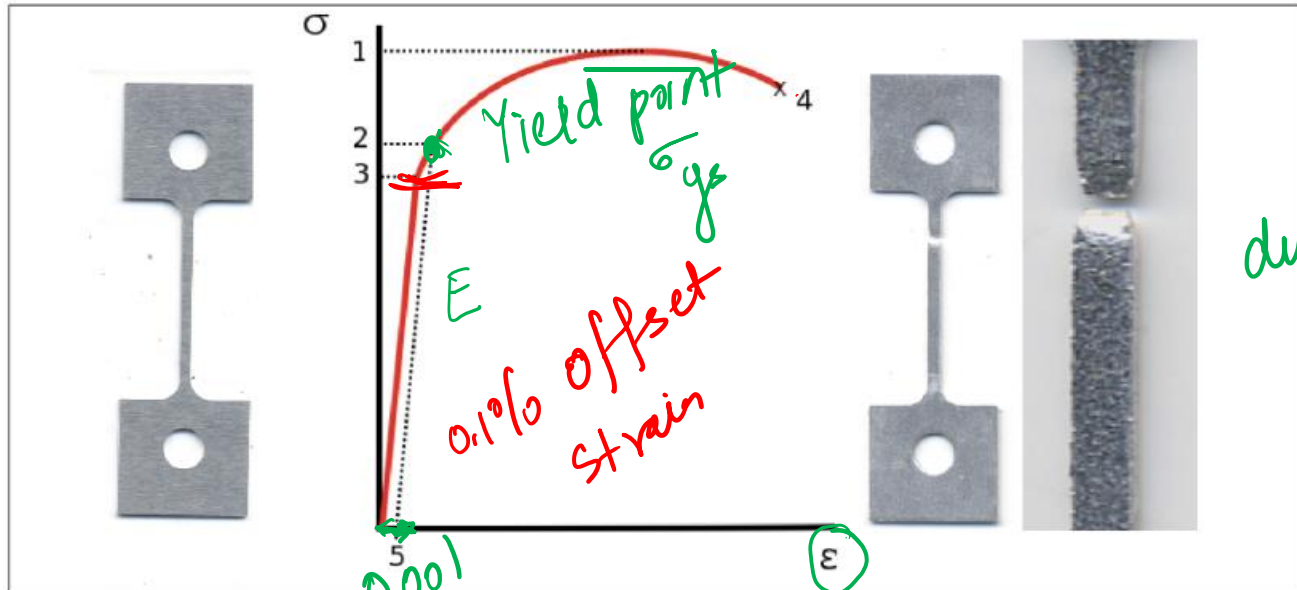
elastic-plastic behavior

Steel  
Fe  $\rightarrow$  BCC alloy



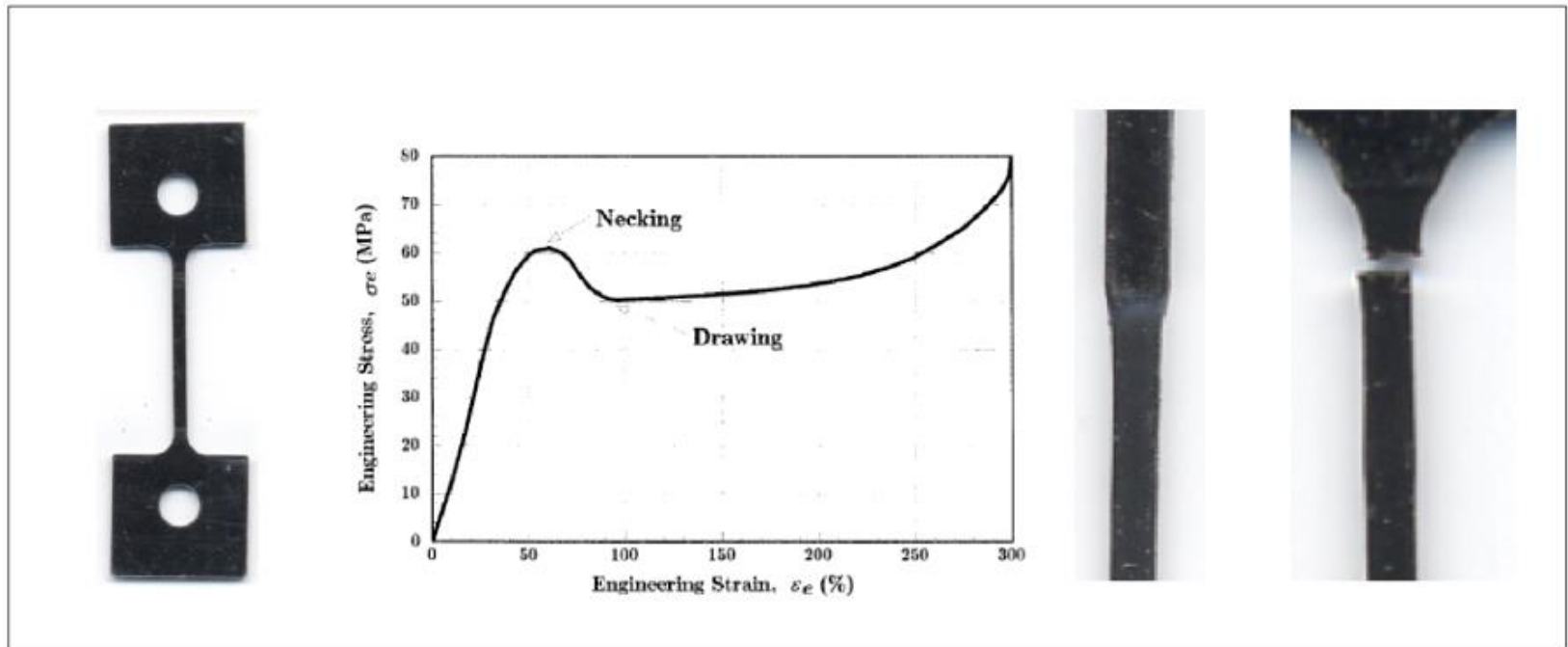
Lower cost rell behavior moderately brittle

Aluminum



ductile

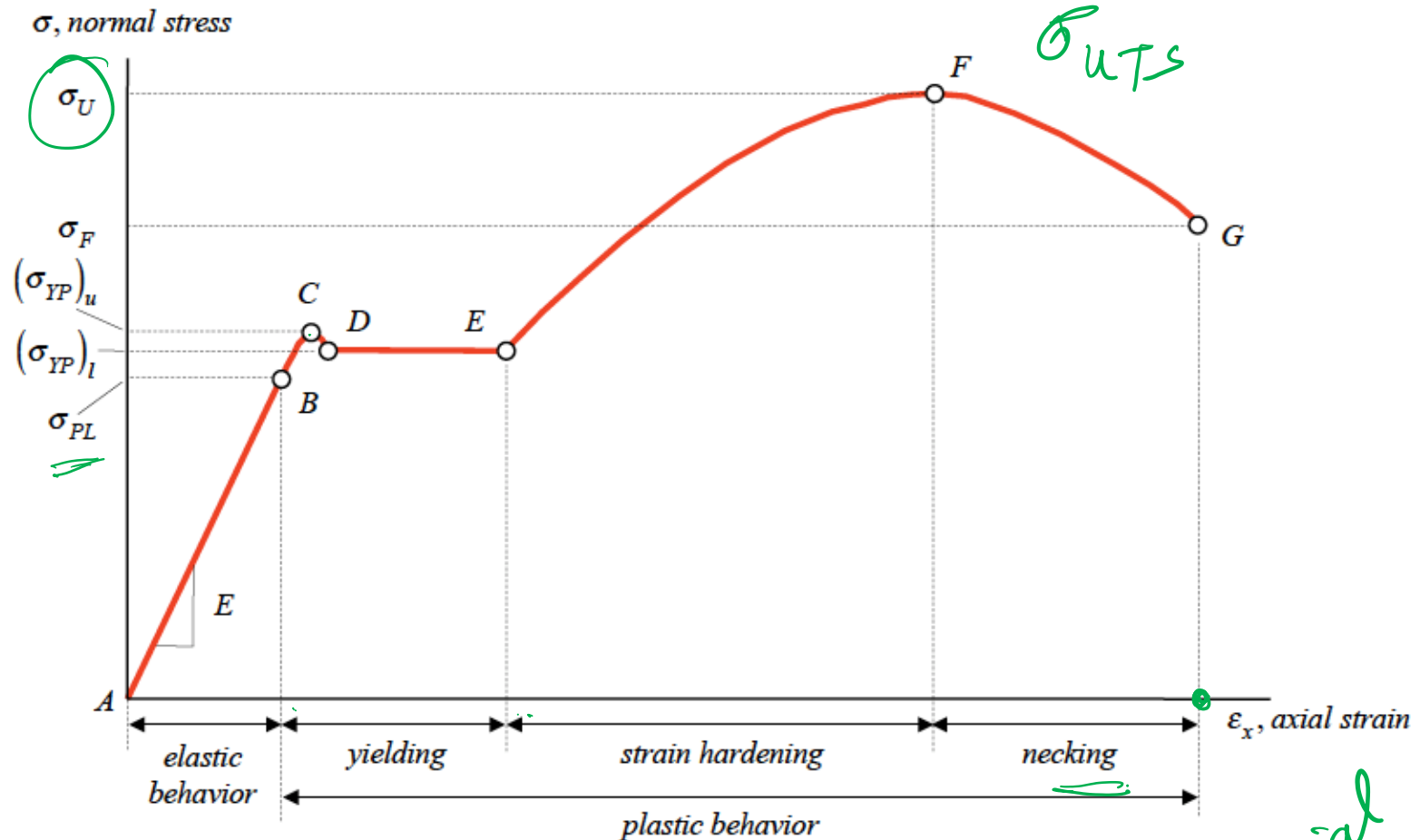
# Stress-Strain Curves



Nylon

→ visco elastic behavior

# Stress-Strain Curves (Steel)



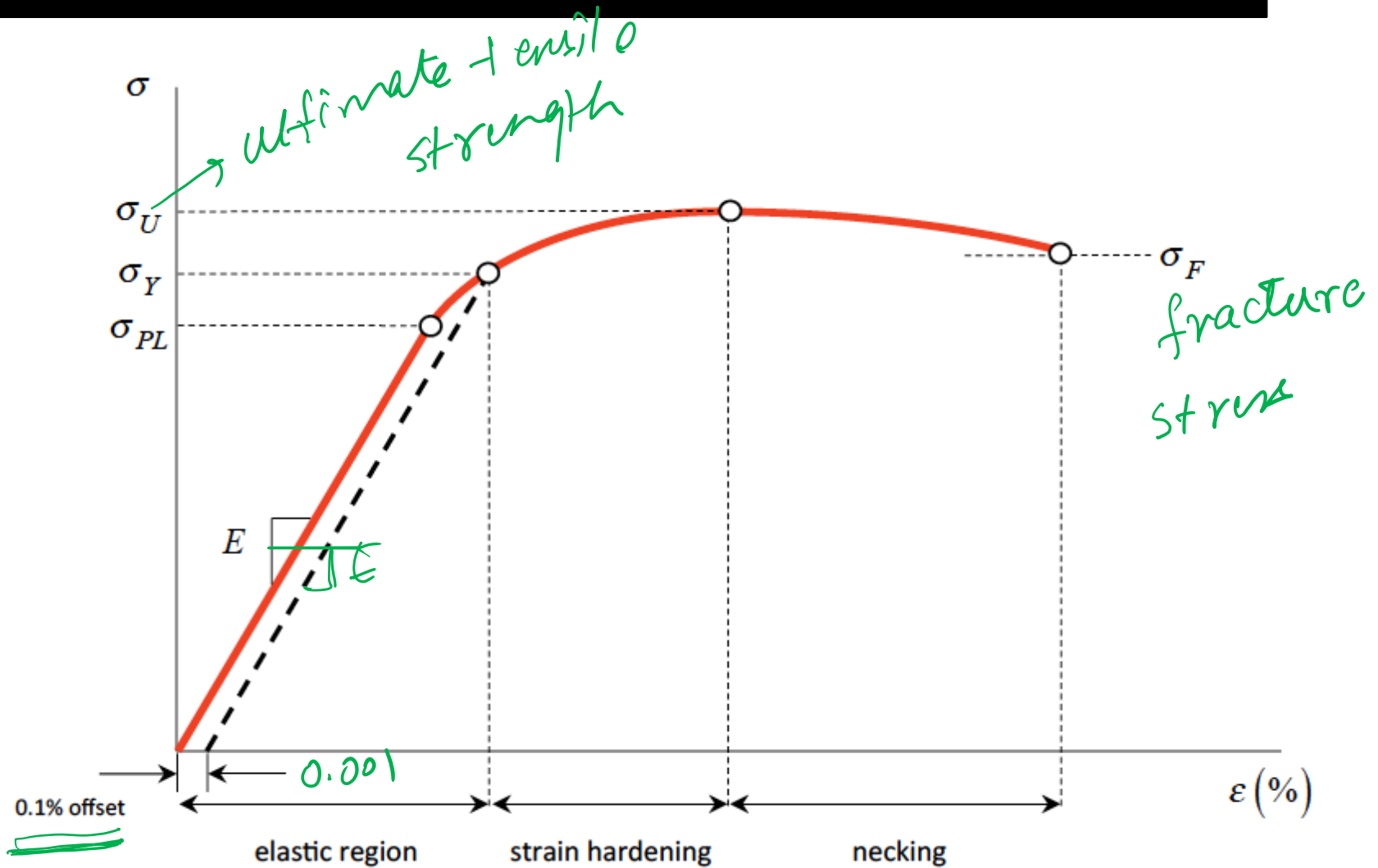
$(\sigma_{YP})_u$  = upper yield point

$(\sigma_{YP})_l$  = lower yield point

$(\sigma_{PL})$  = proportionality limit

material starts to soften

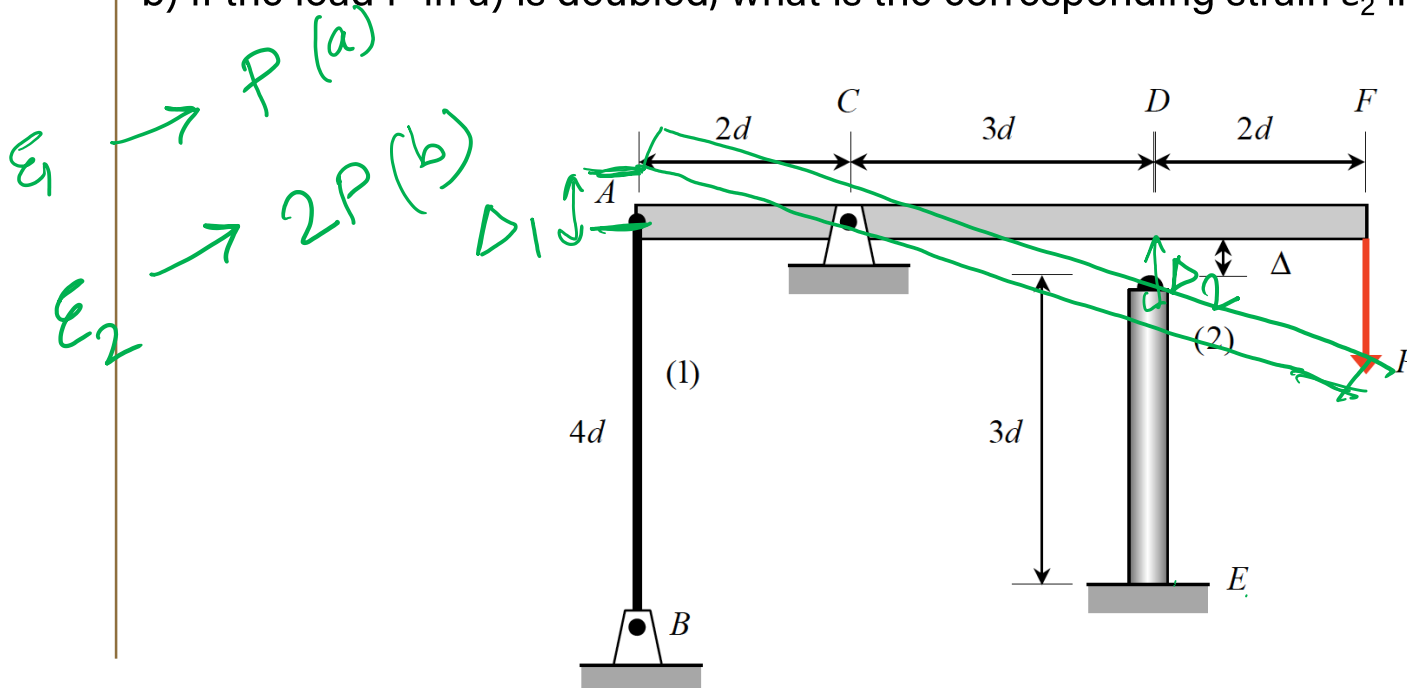
# Stress-Strain Curves (Aluminum)



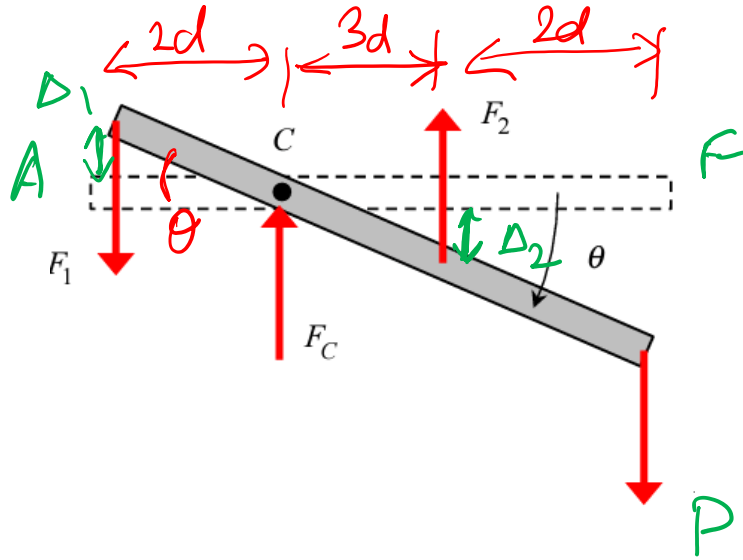
# Example 2.2 from Lecture Book

For small loads  $P$ , the rotation of the rigid beam  $AF$  is controlled by the stretching of rod  $AB$ . For larger loads, the beam comes into contact with the top of column  $DE$ , and further resistance to rotation is shared by the rod and the column. Assume that the clockwise angle  $\theta$  through which beam  $AF$  rotates is small enough to assume that points on the beam essentially move vertically. The cross-sectional areas of members (1) and (2) are  $A_1$  and  $A_2$ , respectively, and the materials of members (1) and (2) have Young's modulus of  $E_1$  and  $E_2$ , respectively.

- A load  $P$  is applied that is just sufficient to close the  $\Delta$  gap between the beam and the column. What is the strain  $\epsilon_1$  in rod  $AB$  for this value of  $P$ ?
- If the load  $P$  in a) is doubled, what is the corresponding strain  $\epsilon_2$  in column  $DE$ ?







$$\tan \theta = \frac{\Delta_1}{2d}$$

$$\tan \theta = \frac{\Delta_2}{3d}$$

$$\frac{\Delta_1}{2d} = \frac{\Delta_2}{3d} \Rightarrow \Delta_1 = \frac{2}{3}(\Delta_2)$$

$$(a) \quad \epsilon_1 = \frac{\Delta_1}{4d} = \frac{2/3 \Delta}{4d} = \frac{\Delta}{6d}$$

$$P_a = ?$$

under equilibrium

$$(\sum M)_C = 0$$

$$F_1(2d) + F_2(3d) - P(5d) = 0$$

$$5P = 2F_1 + 3F_2$$

for (a) member (2) is not compressed  $\Rightarrow F_2 = 0$

$$5P_a = 2F_1 = 2\sigma_1 A_1 \Rightarrow P_a = \frac{2}{5} \sigma_1 A_1$$

$$P_a = \frac{2}{5} E_1 \varepsilon_1 A_1 = \frac{2}{5} E_1 \left( \frac{\Delta}{6d} \right) A_1$$

$$P_a = \frac{1}{15d} (E_1 \Delta A_1)$$

$$\varepsilon_1 = \frac{\Delta}{6d}$$

(b)  $\varepsilon_2 = ?$   $P_b = 2P = \frac{2E_1 \Delta A_1}{15d}$

$$5P = 2F_1 + 3F_2$$

$$5 \left[ \frac{2E_1 \Delta A_1}{15d} \right] = 2E_1 A_1 \varepsilon_1 + 3(E_2 A_2 \varepsilon_2)$$

$$\varepsilon_2 = \Delta_2 - \Delta$$

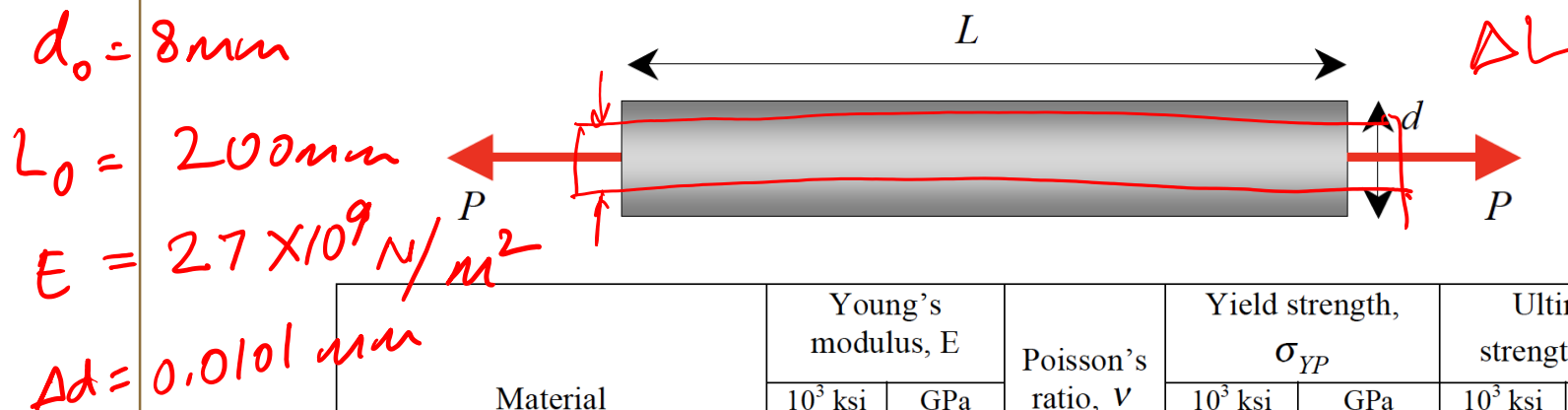
$$P_b = \frac{2E_1 A_1 \Delta}{15d}$$

$$\varepsilon_2 = \frac{E_1 A_1 \Delta}{18d E_2 A_2}$$

# Example 2.4 from Lecture Book

A cylindrical rod having an initial diameter of  $d_0$  and initial length  $L_0$  is made of 6061-T6 aluminum alloy. When a tensile load  $P$  is applied to the rod, its diameter is decreased by  $\Delta d$ .

- Determine the magnitude of the load  $P$ .
- Determine the elongation of the rod over the length of the rod.



Material	Young's modulus, E		Poisson's ratio, $\nu$	Yield strength, $\sigma_{YP}$		Ultimate strength, $\sigma_U$	
	$10^3 \text{ ksi}$	GPa		$10^3 \text{ ksi}$	GPa	$10^3 \text{ ksi}$	GPa
Aluminum alloy 2014-T6	10.6	73	0.33	60	410	70	480
Aluminum alloy 6061-T6	10.0	70	0.33	40	275	45	310
Brass, cold-rolled	15	100	0.34	60	410	75	520
Brass, annealed	15	100	0.34	15	100	40	275
Cast iron, gray	10	70	0.22	-	-	25	170
Steel, ASTM-A36 structural	29	200	0.29	36	250	58	400
Steel, AISI 302 stainless	29	195	0.30	75	520	125	860
Titanium, alloy	16.5	115	0.33	120	830	130	900
Wood, Douglas Fir	1.75	12	-	-	-	7.5	60
Wood, Southern Pine	1.75	12	-	-	-	8.5	60

$d_0 =$  initial diameter

$$\Delta d = d_1 - d_0$$

$d_1 =$  final diameter

$$\epsilon_y = \frac{\Delta d}{d_0} = \frac{-0.0101 \text{ mm}}{8 \text{ mm}} = \frac{-0.0101}{8} = 0.00382$$

$$\nu = 0.33$$

$$\epsilon_x = -\frac{\epsilon_y}{\nu} = + \frac{0.0101/8}{0.33}$$

$$\sigma_x = E(\epsilon_x) = (27 \times 10^9 \text{ N/m}^2)(0.00382)$$

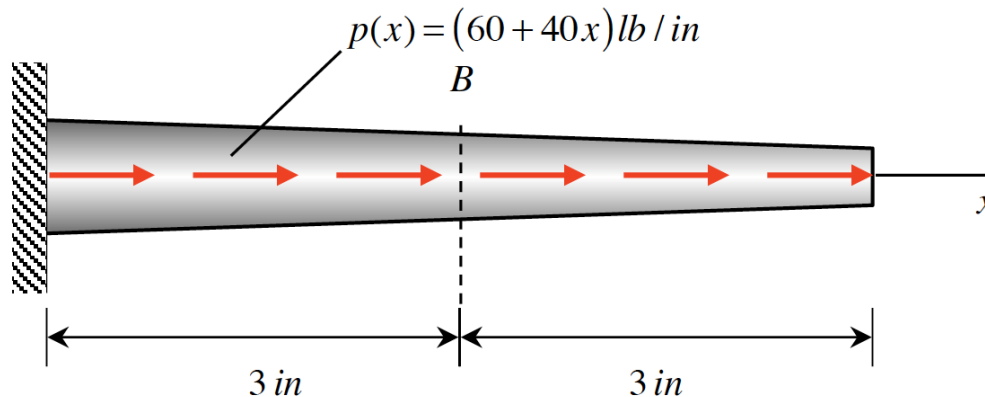
$$\sigma = 0.10314 \times 10^9 \text{ N/m}^2$$

$$P = \sigma A = \sigma \left[ \pi \left( \frac{d_0}{2} \right)^2 \right]$$

$$P = 5182 \text{ N}$$

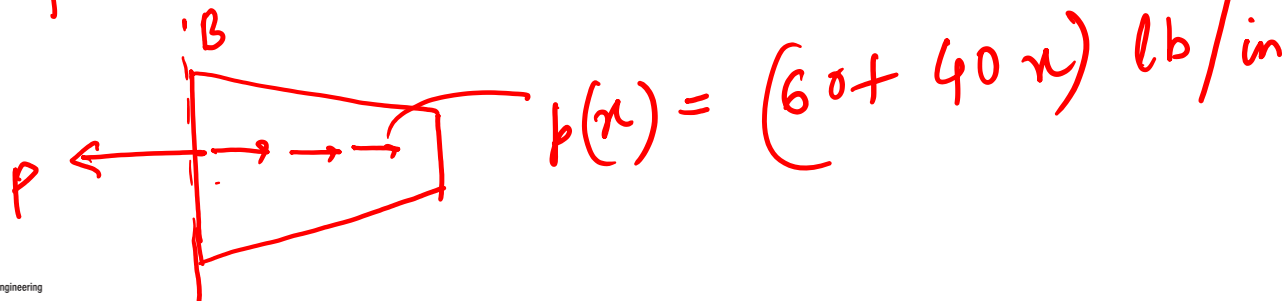
## Example 2.6 from Lecture Book

The tapered rod has a radius of  $r = (2 - x/6)$  in. and is subjected to the distributed loading of  $p = (60 + 40x)$  lb/in. Determine the average normal stress at the center of the rod, B.



$$r(x) = \text{cross-section radius} = \left(2 - \frac{x}{6}\right) \text{ in}$$

Let's first section at B  $\rightarrow$



at equilibrium

$$\sum F_x = 0$$

$$-P \int_{3\text{in}}^{6\text{in}} \left[ (60 + 40x) \frac{\pi b}{\text{in}} \right] dx = 0$$

$$P = \left. 60x + \frac{40x^2}{2} \right|_3^6 = 60(6-3) + 20[6^2 - 3^2]$$
$$= 180 + (20)(36-9)$$
$$= 180 + 540$$

$$P = 720 \text{ lb}$$

$$A = \pi r^2 = \pi \left[ 2 - \frac{x}{6} \right]^2 = \pi \left[ 2 - \frac{3}{6} \right]^2 = 7.07 \text{ in}^2$$

$$\sigma_{ave} = \frac{P}{A} = \frac{720 \text{ lb}}{7.07 \text{ in}^2} = 102 \text{ psi}$$

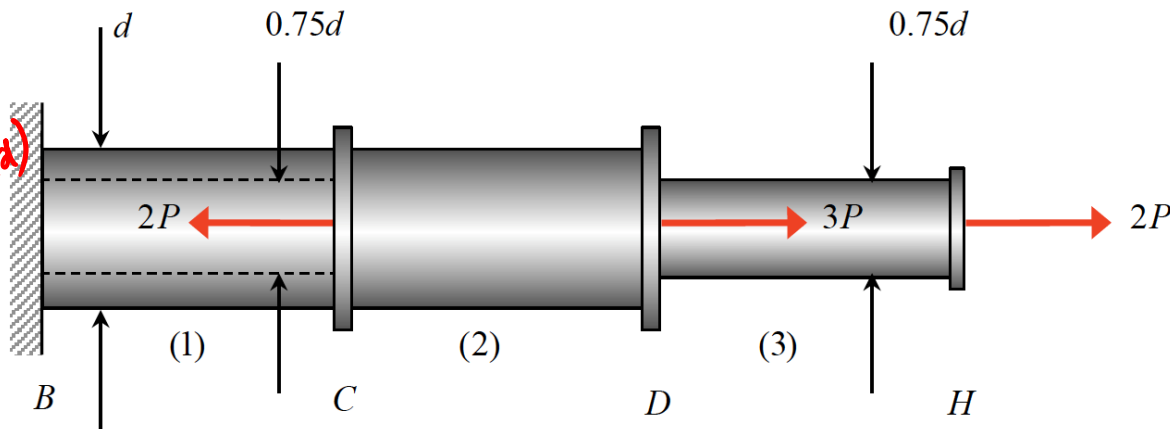
$$\sigma_{ave} = 102 \text{ psi}$$



## Example 2.7 from Lecture Book

The three-segment axially-loaded member shown below is made up of a tubular segment (1) with an outer diameter of  $d$  and inner diameter of  $0.75d$ , a solid segment (2) of outer diameter of  $d$  and another solid segment (3) of outer diameter of  $0.75d$ . A set of axial loads are applied at C, D and H. Determine the axial stresses in the three segments.

- (1) tubular
- (2) solid ( $d$ )
- (3) solid ( $0.75d$ )



outer diameter =  $d = d_o = 1 \text{ inch}$

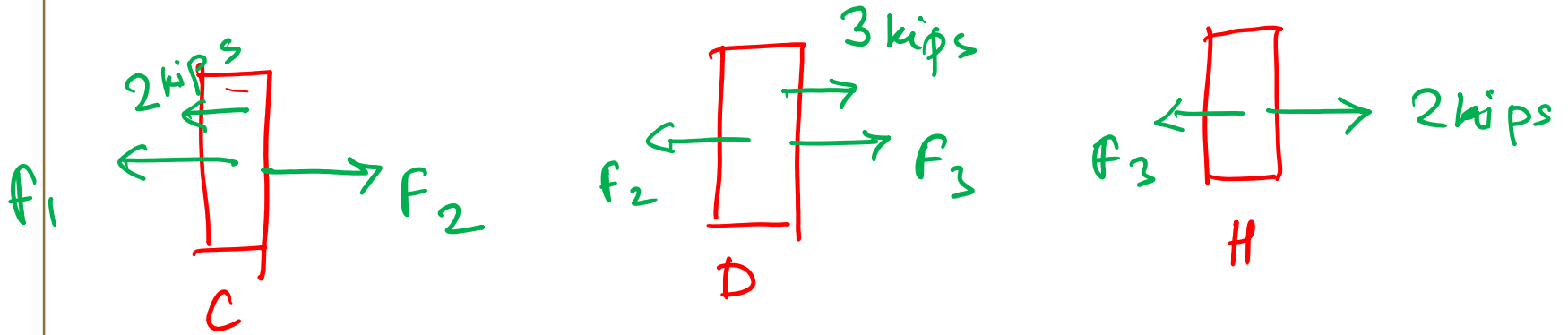
$P = 1 \text{ kips}$

interior dia =  $d_{int} = 0.7 \text{ inch}$

Axial loads at C, B, A are given  
find axial stress for each segment  
(1), (2), (3)

1. free body diagram of connectors

2. Internal resultants = average load in each segment.



at equilibrium  $\Sigma F_x = 0$

for a section

@ H:  $-F_3 + 2 \text{ kips} = 0 \Rightarrow F_3 = 2 \text{ kips}$

@ D

$$F_2 = 3 + F_3 = 5 \text{ kips}$$

@ C

$$F_1 + 2 = F_2$$

$$F_1 = 3 \text{ kips}$$

now we know loads in each section. And they are all tensile.

$$\sigma_1 = \frac{F_1}{A_1} = \frac{3 \text{ kips}}{\pi \left(\frac{d_o}{2}\right)^2 - \pi \left(\frac{d_i}{2}\right)^2}$$

$$= \frac{3 \text{ kips}}{\pi \left(\frac{1}{2}\right)^2 - \pi \left(\frac{0.75}{2}\right)^2}$$

$$\boxed{\sigma_1 = 8.735 \text{ kips/in}^2}$$

$$\sigma_2 = F_2 / A_2 = \frac{5 \text{ kips}}{\pi \left( \frac{d_o}{2} \right)^2} = \frac{5 \text{ kips}}{\pi \left( \frac{1}{2} \right)^2}$$

$$\sigma_2 = 6.37 \text{ kips/in}^2$$

$$\sigma_3 = \frac{F_3}{A_3} = \frac{2 \text{ kips}}{\pi \left( \frac{0.75}{2} \right)^2} = 4.53 \text{ kips/in}^2$$

$$\sigma_3 = 4.53 \text{ kips/in}^2$$

# *THANK YOU*