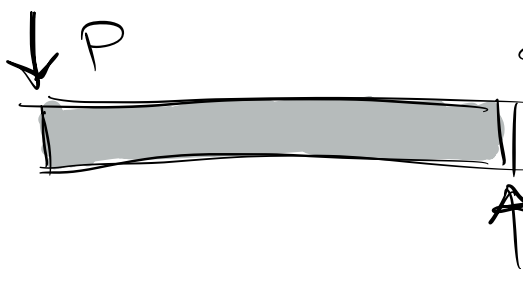


FBD & Equilibrium



$$PL = M_A$$

$$\sum F_y = 0 = R_A - P$$

$$(\sum M) = 0 = PL - M_A$$

Bending moment \Rightarrow normal stress

$$I_{zz} = \frac{(2b)^4}{12} - \frac{b^4}{12} = \frac{15}{12}b^4 = \frac{5}{4}b^4$$

$$\sigma_x > 0$$

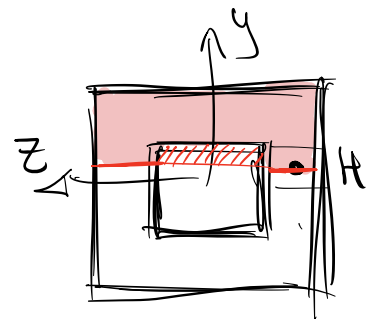


$$\sigma > 0 \text{ (tension) @ } H \Rightarrow \boxed{\sigma = \frac{(PL)(b/4)}{I_{zz}} = \frac{PL}{5b^3}}$$

Shear force \Rightarrow shear stress

$$\tau = \frac{VQ}{I_{zz}t}$$

\Rightarrow Need $Q(y=b/4)$
and $t(y=b/4)$

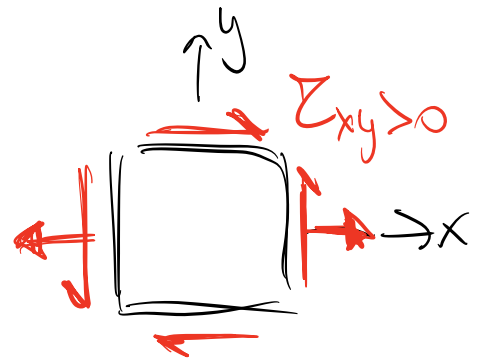
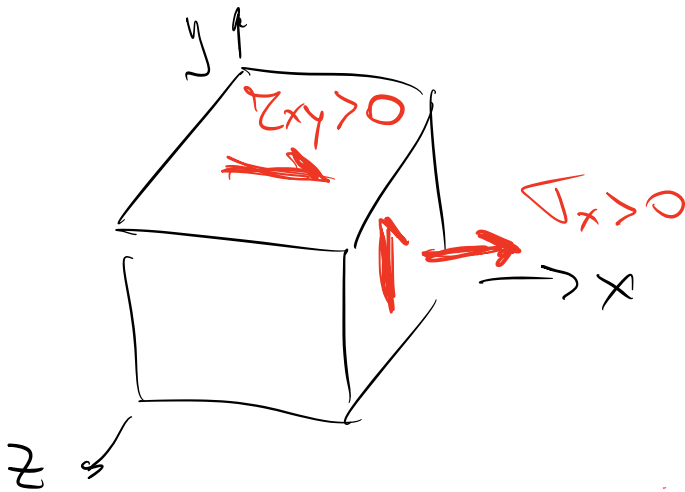


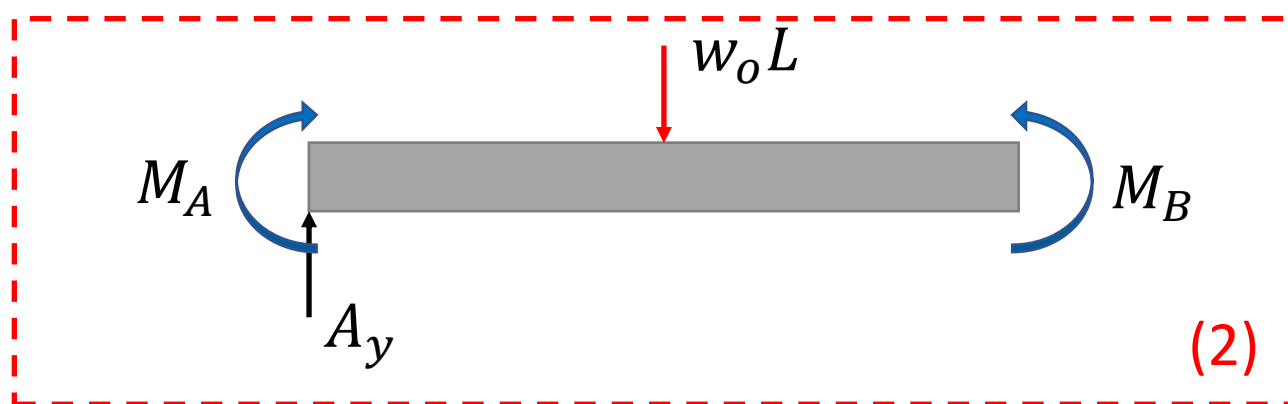
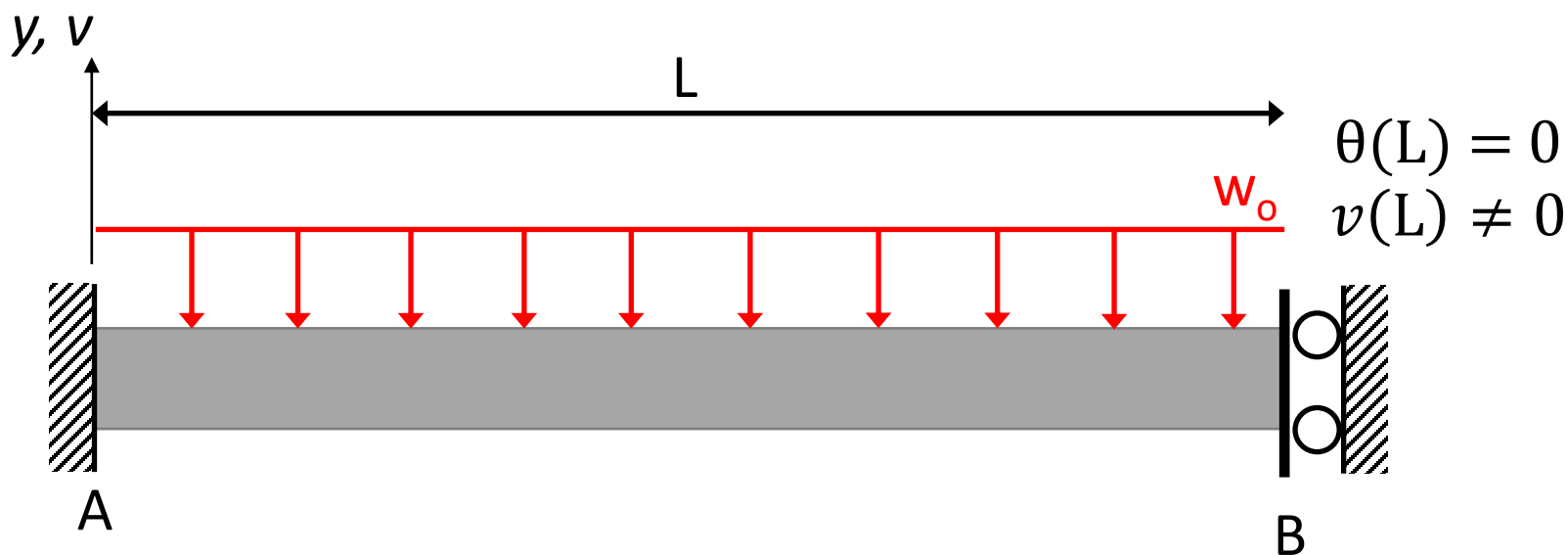
$$Q = \underbrace{\left(2b \times \frac{3}{4}b\right)}_{\Delta_1^*} \underbrace{\left(\frac{3}{8}b + \frac{1}{4}b\right)}_{\gamma_1^*} - \underbrace{\left(b \times \frac{1}{4}b\right)}_{\Delta_2^*} \underbrace{\left(\frac{1}{8}b + \frac{1}{4}b\right)}_{\gamma_2^*} = \frac{27b^3}{32}$$

$$t = \frac{b}{2} + \frac{b}{2} = b$$

$$\Rightarrow \tau = \frac{P \cdot 27b^3/32}{5b^4/4 \times b} = \frac{27P}{40b^2}$$

$\tau_{xy} > 0$
(same direction as shear force)





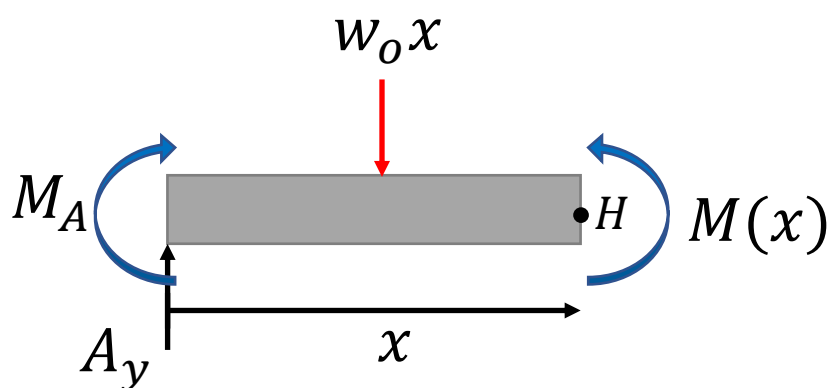
$$\sum F_y = 0 = A_y - w_o L$$

$$A_y = w_o L \quad (1)$$

$$\left(\sum M \right)_A = 0 = -M_A - w_o L \left(\frac{L}{2} \right) + M_B$$

$$M_A = M_B - w_o L \left(\frac{L}{2} \right) \quad (1)$$

\Rightarrow indeterminate



$$\left(\sum M \right)_H = 0 = -M_A - A_y x + w_o x \left(\frac{x}{2} \right) + M(x)$$

$$M(x) = M_A + A_y x - \left(\frac{1}{2} \right) w_o x^2 \quad (3)$$

$$\theta(x) = \theta(0) + \left(\frac{1}{EI}\right) \int_0^x M(x) dx \quad (3)$$

$$\theta(x) = \theta(0) + \left(\frac{1}{EI}\right) \int_0^x M_A + w_o Lx - \left(\frac{1}{2}\right) w_o x^2 dx$$

$$\theta(x) = \theta(0) + \left(\frac{1}{EI}\right) \left[M_A x + \left(\frac{1}{2}\right) w_o Lx^2 - \left(\frac{1}{6}\right) w_o x^3 \right]$$

$$\theta(x) = \theta(0) + \left(\frac{1}{EI}\right) \left[M_A x + \left(\frac{1}{2}\right) w_o Lx^2 - \left(\frac{1}{6}\right) w_o x^3 \right] \quad (1)$$

$$\theta(L) = \left(\frac{1}{EI}\right) \left[M_A L + \left(\frac{1}{2}\right) w_o L^3 - \left(\frac{1}{6}\right) w_o L^3 \right] = 0 \quad (1)$$

$$M_A = -\left(\frac{1}{3}\right) w_o L^2 \quad (1) \quad A_y = w_o L \quad (1)$$

$$M_B = M_A + w_o L \left(\frac{L}{2}\right) \quad M_B = \left(\frac{1}{6}\right) w_o L^2 \quad (1)$$

$$\theta(x) = \left(\frac{1}{EI}\right) \left[M_A x + \left(\frac{1}{2}\right) w_o L x^2 - \left(\frac{1}{6}\right) w_o x^3 \right]$$

$$\theta(x) = \left(\frac{1}{EI}\right) \left[-\left(\frac{1}{3}\right) w_o x L^2 + \left(\frac{1}{2}\right) w_o x^2 L - \left(\frac{1}{6}\right) w_o x^3 \right]$$

$$v(x) = v(0) + \left(\frac{1}{EI}\right) \int_0^x -\left(\frac{1}{3}\right) w_o x L^2 + \left(\frac{1}{2}\right) w_o x^2 L - \left(\frac{1}{6}\right) w_o x^3 dx \quad (3)$$

$$v(x) = v(0) + \left(\frac{1}{EI}\right) \left[-\left(\frac{1}{6}\right) w_o x^2 L^2 + \left(\frac{1}{6}\right) w_o x^3 L - \left(\frac{1}{24}\right) w_o x^4 \right]$$

$$v(x) = v(0) + \left(\frac{1}{EI}\right) \left[-\left(\frac{1}{6}\right) w_o x^2 L^2 + \left(\frac{1}{6}\right) w_o x^3 L - \left(\frac{1}{24}\right) w_o x^4 \right]$$

$$v(x) = \overset{0}{\cancel{v(0)}} + \left(\frac{1}{EI}\right) \left[-\left(\frac{1}{6}\right) w_o x^2 L^2 + \left(\frac{1}{6}\right) w_o x^3 L - \left(\frac{1}{24}\right) w_o x^4 \right] \quad (1)$$

$$v(x) = \left(\frac{1}{EI}\right) \left[-\left(\frac{1}{6}\right) w_o x^2 L^2 + \left(\frac{1}{6}\right) w_o x^3 L - \left(\frac{1}{24}\right) w_o x^4 \right] \quad (2)$$

$$v(L) = \left(\frac{1}{EI}\right) \left[-\left(\frac{1}{6}\right) w_o L^4 + \left(\frac{1}{6}\right) w_o L^4 - \left(\frac{1}{24}\right) w_o L^4 \right]$$

$$v(L) = \left(\frac{1}{EI}\right) \left[-\left(\frac{1}{24}\right) w_o L^4 \right] \quad (1)$$

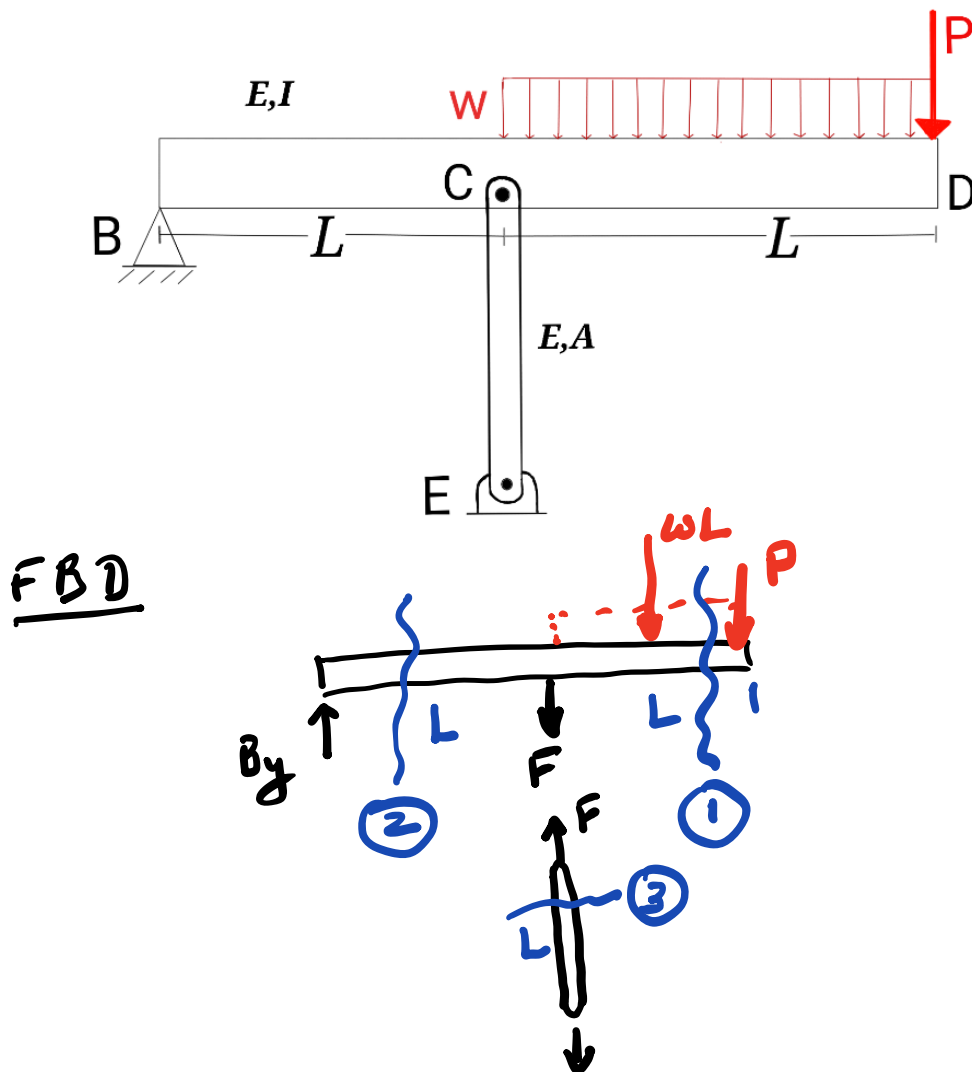


Name (Print) SOLUTION
 (Last) (First)

PROBLEM # 3 (25 points)

A beam BCD of length $2L$ is pinned at B and is connected to a member CE at a distance L from B. The length of CE is L . The beam and the member are all made of the same material with Young's modulus, E . The beam is uniform with a second moment of the area of I , while the cross-sectional area of CE is A . The beam is subjected to a uniform distributed load of w between C and D and a concentrated force of P .

Use Castigliano's second theorem to calculate the vertical deflection of the beam at D. **Neglect the strain energy due to shear forces.**





Name (Print) _____
(Last) (First)

PROBLEM # 3 (cont.)

Equilibrium eqn's

$$\sum M_B = -FL - \omega L\left(\frac{3L}{2}\right) - P(2L) = 0$$

$$F = -2P - \frac{3}{2}\omega L \quad (1)$$

$$\sum F_y = B_y - F - \omega L - P = 0$$

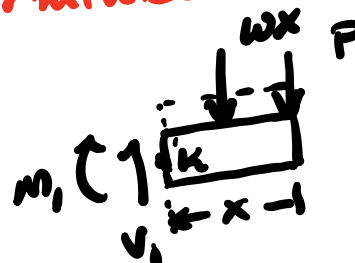
$$B_y = F + \omega L + P = -2P - \frac{3}{2}\omega L + \omega L + P$$

$$B_y = -P - \frac{\omega L}{2} \quad (2)$$

Statically determinate

3 sections:

Section CD:



$$\sum M_k = -m_1 - \omega x\left(\frac{x}{2}\right) - Px = 0$$

$$m_1(x) = -Px - \frac{\omega x^2}{2}$$

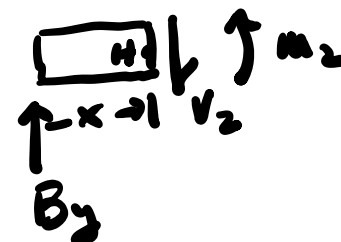
Section BC

$$\sum M_H = m_2 - B_y x = 0$$

$$m_2 = B_y x$$

Replace B_y from (2)

$$m_2(x) = \left(-P - \frac{\omega L}{2}\right)x$$





Name (Print) _____
(Last) (First)

PROBLEM # 3 (cont.)

Member CE is loaded axially with
constant force F
Strain energy: $U = U_1 + U_2 + U_3$

$$U_1 = \frac{1}{2EI} \int_0^L m_1^2 dx$$

$$U_2 = \frac{1}{2EI} \int_0^L m_2^2 dx$$

$$U_3 = \frac{1}{2EA} \int_0^L F^2 dy$$

Castigliano's 2nd theorem: Deflection at D:

$$v_D = \frac{\partial U}{\partial P} = \frac{1}{EI} \int_0^L m_1 \frac{\partial m_1}{\partial P} dx$$

$$+ \frac{1}{EI} \int_0^L m_2 \frac{\partial m_2}{\partial P} dx$$

$$+ \frac{1}{EA} \int_0^L F \frac{\partial F}{\partial P} dy$$

$$\frac{\partial m_1}{\partial P} = -x \quad \frac{\partial m_2}{\partial P} = -x \quad \frac{\partial F}{\partial P} = -2$$

$$\begin{aligned}
 v_D &= \frac{1}{EI} \int_0^L \left(-Px - \frac{\omega x^2}{2} \right) (-x) dx \\
 &+ \frac{1}{EI} \int_0^L \left(-P - \frac{\omega L}{2} \right) x (-x) dx \\
 &+ \frac{1}{EA} \int_0^L \left(-2P - \frac{3}{2} \omega L \right) (-2) dy
 \end{aligned}$$

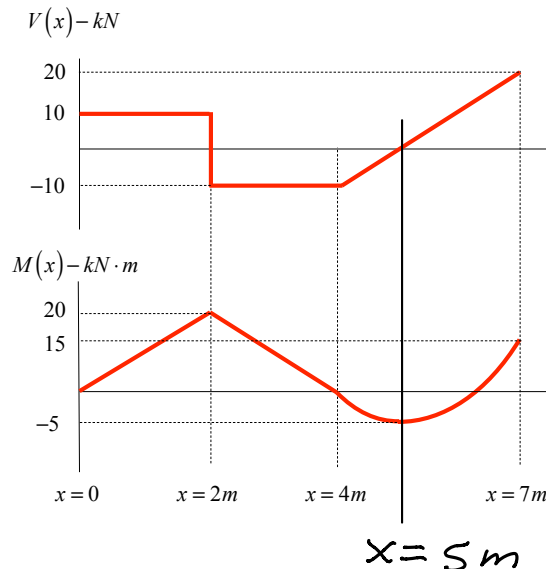
$$\begin{aligned}
 v_D &= \frac{1}{EI} \left[P \frac{x^3}{3} + \frac{\omega x^4}{8} \right]_0^L \\
 &+ \frac{1}{EI} \left[P \frac{x^3}{3} + \frac{\omega L x^3}{6} \right]_0^L \\
 &+ \frac{1}{EA} \left[(4P + 3\omega L) y \right]_0^L \\
 &= \frac{1}{EI} \left[P \frac{L^3}{3} + \frac{\omega L^4}{8} + \frac{PL^3}{3} + \frac{\omega L^4}{6} \right] \\
 &+ \frac{L}{EA} [4P + 3\omega L]
 \end{aligned}$$

$$v_D = \frac{1}{EI} \left[\frac{2PL^3}{3} + \frac{7}{24} \omega L^4 \right] + \frac{L}{EA} [4P + 3\omega L]$$

Name (Print) SOLUTION
(Last) (First)

PROBLEM # 4 (25 points total – partial credit will not be granted)

Part 4A



• At $x=2m \nmid y=\frac{h}{3}$:

$$\tau = -\frac{(20)(h/3)}{I} = -\frac{20}{3} \frac{h}{I} \text{ (C)}$$

• At $x=2m \nmid y=-2h/3$:

$$\tau = -\frac{(20)(-2h/3)}{I} = \frac{40}{3} \frac{h}{I} \text{ (T)}$$

• At $x=5m \nmid y=\frac{h}{3}$:

$$\tau = -\frac{(5)(h/3)}{I} = -\frac{5}{3} \frac{h}{I} \text{ (T)}$$

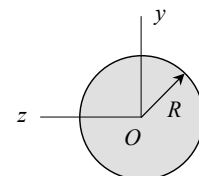
• At $x=5m \nmid y=-\frac{2h}{3}$:

$$\tau = -\frac{(5)(-2h/3)}{I} = \frac{10}{3} \frac{h}{I} \text{ (C)}$$

The shear force/bending moment diagrams for a loaded beam are shown above.

4.A.1 - 2 points: If the beam has the circular cross-section shown to the right, identify the xy -components of the location of a point on the beam having the *maximum magnitude shear stress*.

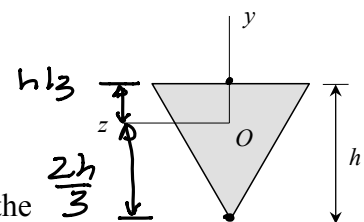
$x = \underline{7m}$ and $y = \underline{0}$



O is the centroid of the cross-section

4.A.2 - 2 points: If the beam has the triangular cross-section shown to the right, identify the xy -components of the location of a point on the beam having the *maximum compressive normal stress*.

$x = \underline{2m}$ and $y = \underline{\frac{h}{3}}$

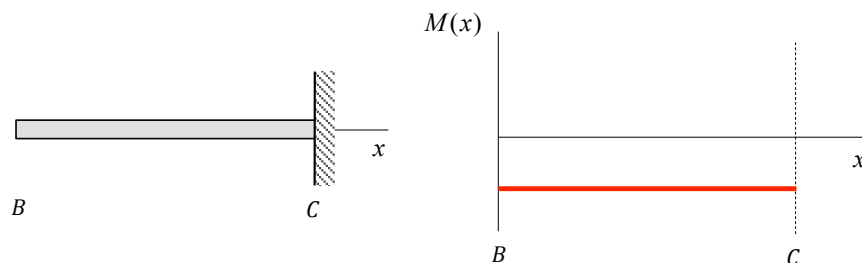


O is the centroid of the cross-section

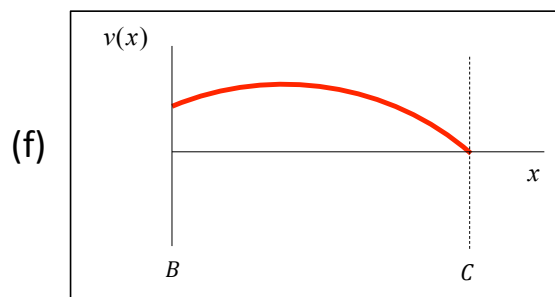
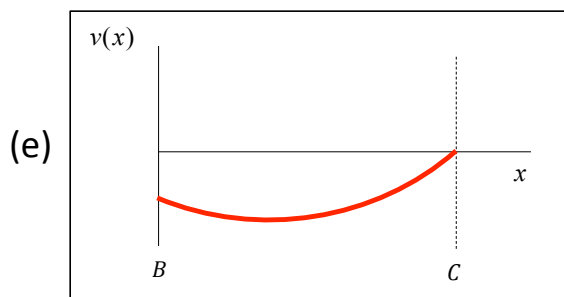
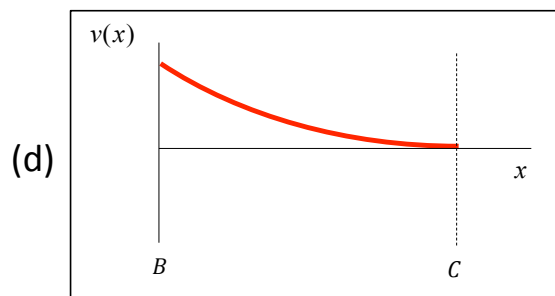
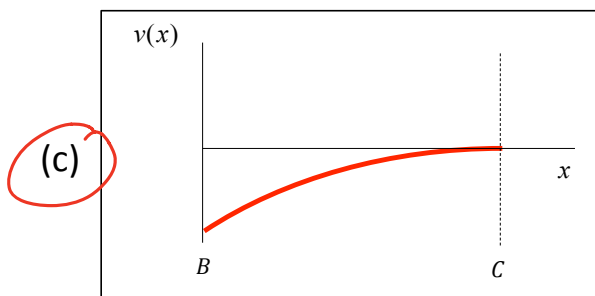
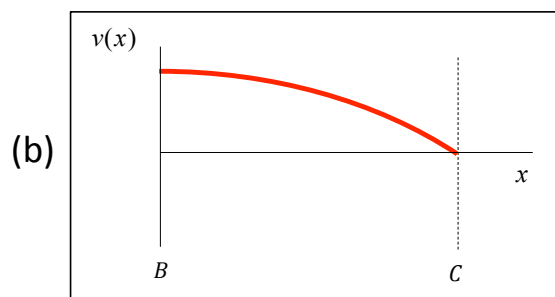
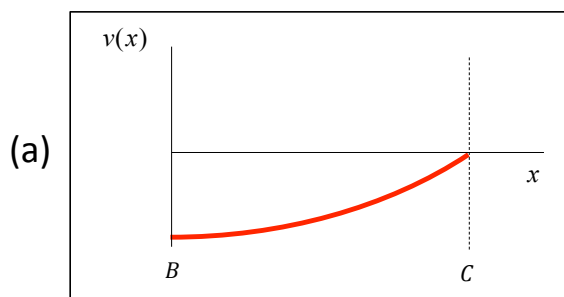
4.A.3 - 2 points: If the beam has the triangular cross-section shown to the right, identify the xy -components of the location of a point on the beam having the *maximum tensile normal stress*.

$x = \underline{2m}$ and $y = \underline{-\frac{2h}{3}}$

Part 4B - 3 points



The bending moment $M(x)$ in a cantilevered beam is shown above. (Note that the actual loading on the beam is not known.) Circle the plot of $v(x)$ below that most closely represents the deflection of this cantilevered beam.

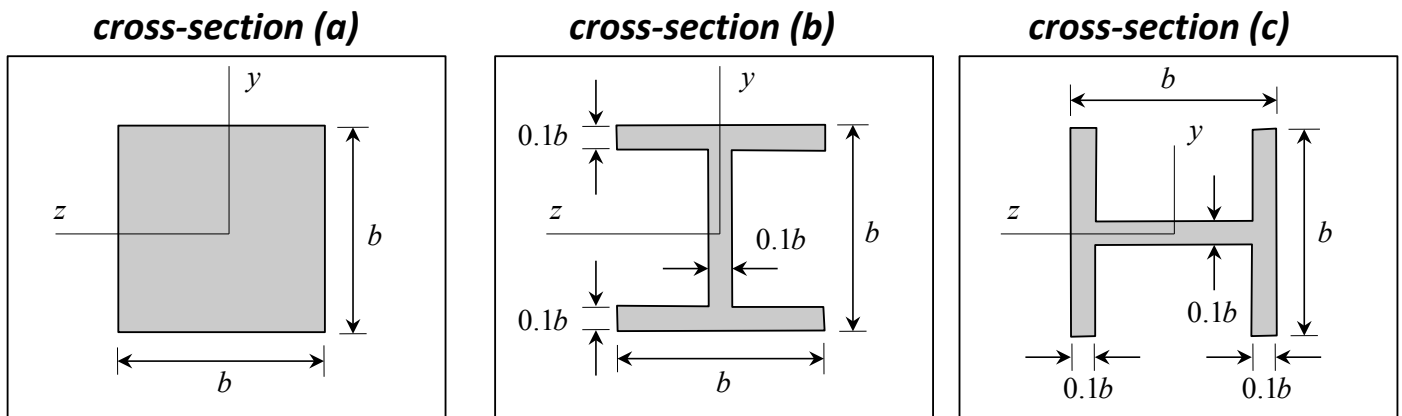


$$\left. \begin{array}{l} V_C = 0 \\ \Theta_C = 0 \end{array} \right\} \Rightarrow \left\{ \frac{d\Theta}{dx} = \frac{1}{EI} M < 0 \text{ over entire length of beam} \right. \\ \Rightarrow \text{(c)}$$

Part 4C



A cantilevered beam is loaded by a concentrated couple M_0 at end C. Consider three possible cross-sections for this beam shown below:



$$I_a > I_b > I_c \quad \& \quad |\sigma| = \frac{M|y|}{I}$$

4.C.1 - 2 points: Which cross-section above produces the *largest maximum magnitude normal stress* in the beam? Circle your response below.

- a) Cross-section (a)
- b) Cross-section (b)
- ☒ c) Cross-section (c)
- d) All three cross-sections produce the same maximum magnitude normal stress.

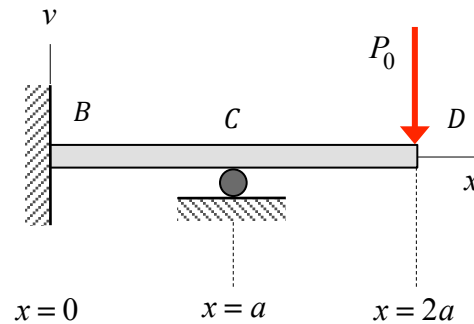
$$I_c \text{ is smallest} \Rightarrow |\sigma| = \text{largest}$$

4.C.2 - 2 points: Which cross-section above produces the *smallest maximum normal stress* in the beam? Circle your response below.

- ☒ a) Cross-section (a)
- b) Cross-section (b)
- c) Cross-section (c)
- d) All three cross-sections produce the same maximum magnitude normal stress.

$$I_a \text{ is largest} \Rightarrow |\sigma| = \text{smallest}$$

Part 4D



A propped-cantilevered beam is loaded with a point load P_0 at end D. Let $V(x)$, $M(x)$ and $v(x)$ represent the shear force, bending moment and deflection, respectively, along the length of the beam.

4.D.1 - 2 points: Which of the following are *zero* at $x=0$? Circle your responses.

$V(0)=0$; $M(0)=0$; $\frac{dv}{dx}(0)=0$; $v(0)=0$;

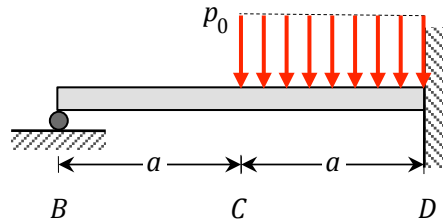
4.D.2 - 2 points: Which of the following are *zero* at $x=a$? Circle your responses.

$V(a)=0$; $M(a)=0$; $\frac{dv}{dx}(a)=0$; $v(a)=0$;

4.D.3 - 2 points: Which of the following are *zero* at $x=2a$? Circle your responses.

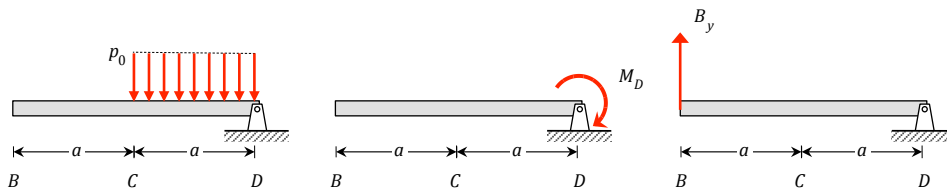
$V(2a)=0$; $M(2a)=0$; $\frac{dv}{dx}(2a)=0$; $v(2a)=0$;

Part 4E



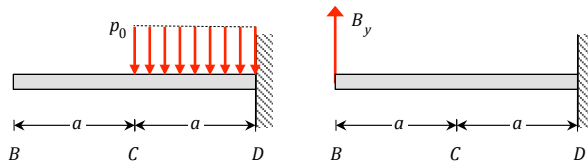
It is desired to determine the reactions acting on the indeterminate beam above using the principle of superposition.

4.E.1 - 1 point: *TRUE* or *FALSE*, the following set of three beam problems can be used with superposition to solve the problem above.

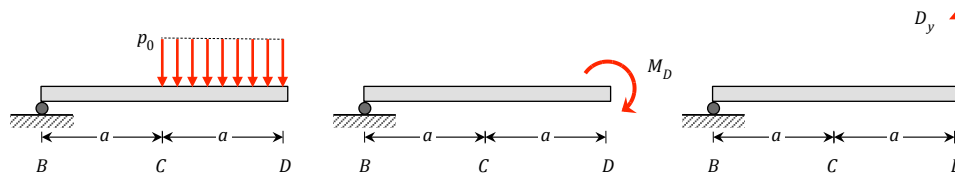


Each problem must be in equilibrium

4.E.2 - 1 point: *TRUE* or *FALSE*, the following set of two beam problems can be used with superposition to solve the problem above.

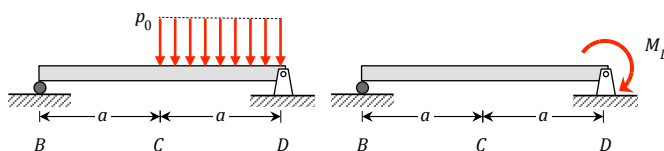


4.E.3 - 1 point: *TRUE* or *FALSE*, the following set of three beam problems can be used with superposition to solve the problem above.

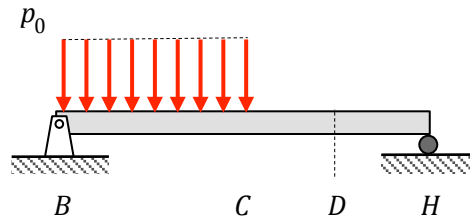


Each problem must be in equilibrium

4.E.4 - 1 point: *TRUE* or *FALSE*, the following set of two beam problems can be used with superposition to solve the problem above.



Part 4F – 2 point

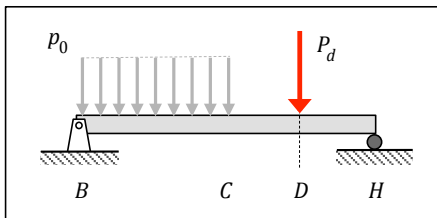


It is desired to use Castigliano's method to determine the vertical displacement of the beam at location D as a result of the applied loading shown. With this approach, a strain energy function for the beam needs to be developed. Which of the loading systems below is needed for establishing this strain energy?

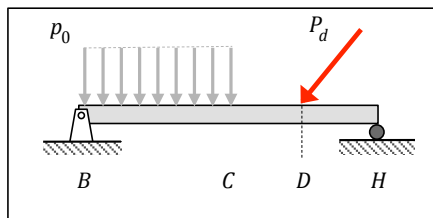
- a) loading system (a)
- b) loading system (b)
- c) loading system (c)
- d) the original loading system above

Dummy force must be in the direction of desired displacement.

loading system (a)



loading system (b)



loading system (c)

